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Numerical determination and treatment of convective heat transfer coefficient in the coupled building energy and CFD simulation

Zhiqiang Zhai^{a,*}, Qingyan (Yan) Chen^b

^aDepartment of Civil, Environmental and Architectural Engineering, University of Colorado, UCB 428, ECOT 441, Boulder CO 80309-0428, USA ^bRay W. Herrick Laboratories, School of Mechanical Engineering, Purdue University, 585 Purdue Mall, West Lafayette IN 47907-2088, USA

Abstract

The integration of building energy simulation (ES) and computational fluid dynamics (CFD) programs eliminates many assumptions employed in the separate applications, resulting in more accurate predictions of building performance. This paper discusses the methods used to determine convective heat transfer on interior surfaces of building envelope, which is the key linkage between ES with CFD programs. The study found that the size of the first grid near a wall in CFD is crucial for the correct prediction of the convective heat. A finer grid resolution in CFD does not always lead to a more accurate solution when using zero-equation turbulence models. Through numerical experiments, the paper suggests a universal first grid size at 0.005 m for natural convection room airflows and 0.1 m for forced convection indoor airflows. The investigation also found that negative convective heat transfer coefficients may cause divergence and instability of the coupled simulation.

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1. Introduction

A building energy simulation (ES) program predicts building thermal performance, while a computational fluid dynamics (CFD) program calculates detailed airflow characters. Both of them are important tools for building performance evaluation and have been widely used in building and system design practices. Many literatures on this topic are available, such as those by Clarke [1], Chen [2], Martin [3], Chen and Srebic [4], etc. The coupling of ES and CFD can take full advantages of the individual programs and produce more accurate and complete results, because they can provide complementary boundary conditions to each other. For example, ES can provide interior surface temperatures of building envelopes and heating/cooling load to CFD as boundary conditions while CFD can determine accurate surface convective heat fluxes for ES. The benefits of the integration have been discussed in many previous studies, such as, by Clarke et al. [5], Nielsen and Tryggvason [6], Srebric et al. [7], Beausoleil-Morrison et al. [8], and Zhai et al. [9].

In a coupled simulation, ES calculates the heat conduction through building envelopes and CFD determines air movement in an indoor space. The convective heat transfer on interior surfaces of building envelope is the most important linkage to couple these two programs [9]. In the coupling, CFD provides ES convective heat transfer on each enclosure surface. The convective heat transfer calculated by CFD is highly sensitive to the numerical methods and turbulence models employed in CFD. This study analyzes the effect of the size of the first grid to a wall used in CFD on the heat transfer and examines how turbulence models determine the convective heat transfer. In addition, with traditional definition of convective heat transfer coefficient, which is based on the temperature difference of an interior surface and room air, the coefficient value can be negative. The impact of negative coefficient on the coupled simulation is another subject of this study.

2. Factors to numerical solution of convective heat transfer

CFD discretizes the computational domain into many grid cells and solves the governing conservation equations of

^{*} Corresponding author. Fax: +1-303-4927317.

E-mail address: john.zhai@colorado.edu (Z. Zhai).



Fig. 1. Illustration of the grid, cell, node, and distance to the surface in CFD.

flow on these grid cells. As shown in Fig. 1(a), CFD calculates convective heat transfer from a rigid surface through:

$$Q = hA(T_{\text{surface}} - T_1), \tag{1}$$

$$h = C_p \frac{\rho(\nu_t + \nu)}{\Pr} \frac{1}{D},$$
(2)

where *h* is convective heat transfer coefficient, *A* is surface area, T_{surface} is surface temperature, T_1 is air temperature at the first grid node that is at a normal distance of *D* to the surface, C_p is the specific heat of air, Pr is Prandtl number, ρ is air density, v_t is turbulence viscosity determined by turbulence models at the first grid node, and v is the molecular viscosity of air. According to Eqs. (1) and (2), the convective heat transfer is determined by the convective heat transfer coefficient and temperature difference between the surface and air at the first grid node, and the convective heat transfer coefficient is determined by effective viscosity of air (turbulence viscosity plus molecular viscosity) at the first grid node. In order to obtain accurate surface convective heat transfer, it is important to find its relationship with the *D* and turbulence model.

Since the size of the first grids in CFD can be adjusted according to the resolution requirement, the convective heat calculated in this manner could be grid-dependent. Another method is to use a prescribed distance, D_2 , to a wall surface and calculate convective heat transfer based on the air temperature and effective viscosity there, as illustrated in Fig. 1(b). Since the D_2 is a prescribed value, this method would eliminate the grid dependence problem.

However, the convective heat transfer calculated with the second method could be different from that calculated based on the information at the first grid node. For example, if assuming laminar flow and $D_2 = 2D$, the same heat transfer would require $T_{\text{surface}} - T_2 = 2(T_{\text{surface}} - T_1)$. This implies a linear air temperature profile in the region, which is true only at the very close vicinity of the surface for laminar flow. Such a condition is difficult to meet for most indoor airflows.

Therefore, the calculation of convective heat transfer should use the first method. The question now is how to avoid the grid dependence problem or what the size of first grids should be for the correct prediction of convective heat transfer. CFD theory indicates that finer grids provide more accurate results [10]. This may not be true when using simple turbulence models. This paper discusses the grid dependence problem in the following sections by using laminar and turbulent flows over a flat plate as examples.

3. Convective heat transfer in laminar flows

First this investigation considers a laminar flow of forced convection along a horizontal plate and that of natural convection along a vertical plate, as illustrated in Fig. 2. The convective heat transfer computed by CFD is compared with the analytical and empirical solutions to identify the impact of the size of first grid on the heat transfer.

The convective heat transfer through the thermal boundary layer of the plate is

$$q_{\text{surface}} = -k \frac{\partial T}{\partial y} \bigg|_{y=0} = h(T_{\text{surface}} - T_{\infty}),$$
(3)

where k is the fluid conductivity, T_{surface} is the plate surface temperature, T_{∞} is the temperature of the free stream outside of the thermal boundary layer. Eq. (3) shows that the heat conduction is the same as the convective heat transfer, because there is no fluid motion in the direction of heat transfer.

3.1. Forced convection

The exact solution of the laminar plate flow of forced convection is [11]

$$\frac{U}{U_{\infty}} = \frac{T_{\text{surface}} - T}{T_{\text{surface}} - T_{\infty}} = 1 - \frac{T - T_{\infty}}{T_{\text{surface}} - T_{\infty}},\tag{4}$$

$$q_{\text{surface}} = -k\frac{\partial T}{\partial y} = -\rho C_p \frac{v}{\Pr} \frac{\partial T}{\partial y} = \frac{3}{2} k \frac{T_{\text{surface}} - T_{\infty}}{\delta_{\text{t}}}, \quad (5)$$

where δ_t is the thickness of thermal boundary layer and $k = \rho C_p v/\text{Pr}$.

The analytical solution of the boundary layer equations shows $\delta = \delta_t$ when Pr=1, where δ is the thickness of velocity boundary layer. The exact solution of the boundary layer equations produces [11]

$$\delta = 4.92 x / \sqrt{\mathrm{Re}_x},\tag{6}$$



Fig. 2. Laminar plate flows of forced and natural convection.

where $\text{Re}_x = U_{\infty} x / v$ is the Reynolds number of plate flow. For example, if the velocity is 0.1 m/s over a 5-m long plate,

$$\operatorname{Re}_{L/2} = \frac{UL/2}{v} = \frac{0.1 \times 2.5}{1.5 \times 10^{-5}} = 1.7 \times 10^4,$$
(7)

the δ_t at the middle length of the plate is $\delta_t = \delta \approx 0.09$ m.

A CFD program would calculate the convective heat transfer from the plate as

$$q_{\rm CFD} = -\rho C_p \frac{v}{\Pr} \frac{T_{\rm D} - T_{\rm surface}}{D} = k \frac{T_{\rm surface} - T_D}{D},\tag{8}$$

where *D* is the normal distance from the center of the first grid cell to the surface (half of the cell size) and T_D is the air temperature at cell center. In order to analyze possible errors associated with grid sizes, this study considers two different grid sizes: *D* is smaller than δ_t and *D* is larger than δ_t . • $D < \delta_t$.

When D is smaller than δ_t , the possible numerical error is

$$\Delta q = q_{\rm CFD} - q_{\rm surface} = -\rho C_p \frac{v}{\Pr} \frac{T_D - T_{\rm surface}}{D} - q_{\rm surface}$$
$$= k \frac{T_{\rm surface} - T_D}{D} - k \frac{T_{\rm surface} - T_{\infty}}{2/3\delta_{\rm t}}.$$
(9)

Because the temperature profile in the thermal boundary can be approximated as [11]

$$\frac{T_D - T_{\text{surface}}}{T_{\infty} - T_{\text{surface}}} = \frac{3}{2} \frac{D}{\delta_{\text{t}}} - \frac{1}{2} \left(\frac{D}{\delta_{\text{t}}}\right)^3,\tag{10}$$

the relative error of convective heat transfer due to D then becomes

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{T_{\text{surface}} - T_{D}}{(T_{\text{surface}} - T_{\infty})D} - \frac{T_{\text{surface}} - T_{\infty}}{(T_{\text{surface}} - T_{\infty})2/3\delta_{\text{t}}} = \left(\frac{3}{2}\frac{D}{\delta_{\text{t}}} - \frac{1}{2}\left(\frac{D}{\delta_{\text{t}}}\right)^{3}\right)\frac{1}{D} - \frac{1}{2/3\delta_{\text{t}}} = -\frac{1}{2}\frac{D^{2}}{\delta_{\text{t}}^{3}}.$$
(11)

Eq. (11) verifies that the smaller the *D* the more accurate the calculated convective heat transfer. The error of convective heat transfer due to *D* is on the order of $O(D^2)$ for this case. • $D \ge \delta_t$.

The same analysis can be conducted when D is equal to or larger than δ_t , where $T_D = T_\infty$. Then,

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{1}{D} - \frac{1}{2/3\delta_{\text{t}}} = \frac{1}{D} - \frac{3}{2\delta_{\text{t}}}.$$
 (12)

Eq. (12) shows that the minimum error of the calculated heat flux is one-third of the analytical solution in Eq. (5) as *D* equals δ_t . The convective heat transfer calculated will deviate significantly from the actual solution if the size of first grids is unreasonably large.

3.2. Natural convection

For the natural convection case, the analytical solution shows [11]

$$\delta_{t}^{4} = \frac{240(20/21 + \Pr)}{\Pr^{2}g\beta|T_{\text{surface}} - T_{\infty}|/\nu^{2}}x.$$
(13)

With this equation, one can roughly estimate the scale of δ_t due to natural convection in the middle of a 3-m vertical wall at 40°C and room air temperature at 25°C to be 0.024 m.

The analytical solution of the convective heat transfer for natural convection is [11]

$$q_{\text{surface}} = 2 \, \frac{k(T_{\text{surface}} - T_{\infty})}{\delta_{\text{t}}} \tag{14}$$

while the CFD equation for natural convection is the same as Eq. (8) for forced convection. Therefore,

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{T_{\text{surface}} - T_D}{(T_{\text{surface}} - T_{\infty})D} - \frac{2}{\delta_t}.$$
 (15)

• $D < \delta_t$.

When D is smaller than δ_t ,

$$\frac{T_{\text{surface}} - T_D}{T_{\text{surface}} - T_\infty} = 1 - \frac{T_D - T_\infty}{T_{\text{surface}} - T_\infty} = 1 - \left(1 - \frac{D}{\delta_t}\right)^2, \quad (16)$$

$$\frac{\Delta q}{k(T_{\text{surface}} - T_\infty)} = \frac{T_{\text{surface}} - T_D}{(T_{\text{surface}} - T_\infty)D} - \frac{2}{\delta_t}$$

$$= \frac{1}{D} \left(1 - \left(1 - \frac{D}{\delta_t}\right)^2\right) - \frac{2}{\delta_t} = \frac{1}{D} \left(\frac{2D}{\delta_t} - \frac{D^2}{\delta_t^2}\right)$$

$$- \frac{2}{\delta_t} = -\frac{D}{\delta_t^2}.$$
(17)

Hence, the smaller the size of first grids, the better the accuracy. The calculation error of the convective heat transfer due to the size of first grids (2D) is on the order of O(D) for natural convection.

•
$$D \ge \delta_t$$
.
Since $T_D = T_\infty$ when $D \ge \delta_t$, Eq. (15) becomes

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{1}{D} - \frac{2}{\delta_{\text{t}}}.$$
(18)

The equation is very similar to Eq. (12) for forced convection, except that the analytical heat flux is different.

4. Convective heat transfer in turbulent flows

The study on laminar flows verifies that finer grid distribution provides more accurate solutions. However, most indoor airflows are turbulent. The analysis for turbulent flows is more complicated than that for laminar flow, because the accuracy of convective heat transfer predicted in turbulent flows depends on both the size of first grids and turbulence model used. This study has examed the accuracy by using three zero-equation turbulence models and different grid resolutions for a forced convection flow along a plate. The zero-equation turbulence models are frequently used in the CFD simulation for building design due to their simpleness and efficiency. The three zero-equation turbulence models tested here are:

(1) Constant viscosity model:

$$v_{\rm t} = 100v.$$
 (19)

(2) Xu's zero-equation model [12]:

$$v_{\rm t} = 0.03874 \left| U_D \right| l, \tag{20}$$

where l is the normal distance to the surface and U_D is the airflow speed at this location.

(3) Prandtl's zero-equation model [13]:

$$v_{\rm t} = 2\kappa^2 y^2 \left| \frac{\partial U}{\partial y} \right|,\tag{21}$$

where $\kappa = 0.41$ and y is the distance to the surface.

The convective heat transfer in turbulent flows is related to turbulence as defined by

$$q = -\rho C_p \left(\frac{v}{\Pr} + \frac{v_t}{\Pr_t} \right) \frac{\partial T}{\partial y} \Big|_{y=0} = h \left(T_{\text{surface}} - T_{\infty} \right), \quad (22)$$

where v_t is turbulence eddy viscosity and *T* represents Reynolds-averaged temperature. In Eq. (22), $v_t \ge v$ for the flow region away from walls, and $v_t \le v$ for the near wall region, which is called viscous sub-layer. It is obvious that if the first grid is located in the viscous sub-layer, the relationship between the convective heat transfer predicted and the size of first grids is similar to that for laminar flow. The heat transfer computed does not directly depend on the turbulence model, although the turbulence model does influence the velocity and temperature profiles. According to the empirical equation [11],

$$\delta_{\text{turbulence}} = 0.057[(n+1)(n+2)/n]^{0.8} \text{Re}_x^{-0.2} x, \qquad (23)$$

where n = 7 for a common velocity profile in the boundary layer, $\delta_{turbulence}$ for a forced convection with a velocity 0.1 m/s over a 5-m long plate is

$$\delta_{\text{turbulence}} = 0.13 \text{ m} \tag{24}$$

and $\delta_{\text{sub-layer}} \approx 0.15 - 0.2 \delta_{\text{turbulence}} = 0.02 - 0.03 \text{ m}$. When the velocity is increased to 1 m/s, $\delta_{\text{turbulence}}$ and δ_{sub} -layer are reduced to 0.08 and 0.01-0.02 m, respectively.

The empirical solution for the turbulent plate flow and heat transfer gives [11]

$$Nu_x = 0.0296 Re_x^{0.8} Pr^{1/3} = hx/k.$$
 (25)

Considering the heat transfer in the middle of the plate with a velocity of 0.1 m/s,

$$Nu_{L/2} = 0.0296 Re_{L/2}^{0.8} Pr^{1/3} = 0.0296 \times \left(\frac{0.1 \times 2.5}{1.5 \times 10^{-5}}\right)^{0.8} \times 1^{1/3} = 70.6 = hL/2k.$$
(26)

As a result,

$$q_{\text{surface}} = h(T_{\text{surface}} - T_{\infty}) = 28.2k(T_{\text{surface}} - T_{\infty}).$$
(27)

On the other hand, with the assumption of $Pr_t = Pr$, a CFD program would calculate the heat transfer as

$$q_{\rm CFD} = -\rho C_p \left(\frac{\nu}{\Pr} + \frac{\nu_{\rm t}}{\Pr_{\rm t}} \right) \frac{\partial T}{\partial y} \Big|_{y=0}$$
$$= -\rho C_p \frac{\nu + \nu_{\rm t}}{\Pr} \frac{T_D - T_{\rm surface}}{D}$$
$$= -k \frac{\nu + \nu_{\rm t}}{\nu} \frac{T_D - T_{\rm surface}}{D}.$$
(28)

•
$$D \ge \delta_t$$
.
Since $T_D = T_\infty$ as $D \ge \delta_t$,

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{q_{\text{CFD}} - q_{\text{surface}}}{k(T_{\text{surface}} - T_{\infty})}$$
$$= \frac{v + v_{\text{t}}}{v} \frac{1}{D} - 28.2.$$
(29)

With constant viscosity model:

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{v + v_{\text{t}}}{v} \frac{1}{D} - 28.2 = \frac{101}{D} - 28.2.$$
(30)

With Xu's zero-equation model:

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{v + v_{\text{t}}}{v} \frac{1}{D} - 28.2$$
$$\approx \frac{v_{\text{t}}}{v} \frac{1}{D} - 28.2 = 230. \tag{31}$$

With Prandtl's zero-equation model:

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{v + v_t}{v} \frac{1}{D} - 28.2$$
$$\approx \frac{v_t}{v} \frac{1}{D} - 28.2 = 2213. \tag{32}$$

• $D < \delta_{t}$.

The temperature and velocity profiles in the boundary layer are approximated as [11]

$$\frac{T - T_{\text{surface}}}{T_{\infty} - T_{\text{surface}}} = \frac{U}{U_{\infty}} = \left(\frac{y}{\delta}\right)^{1/n}.$$
(33)

Hence,

$$\frac{T_D - T_{\text{surface}}}{T_{\infty} - T_{\text{surface}}} = \frac{U_D}{U_{\infty}} = \left(\frac{D}{\delta}\right)^{1/n}$$
(34)

and

$$q_{\rm CFD} = -k \frac{v + v_{\rm t}}{v} \frac{T_D - T_{\rm surface}}{D}$$
$$= -k \frac{v + v_{\rm t}}{v} \frac{T_\infty - T_{\rm surface}}{D} \left(\frac{D}{\delta}\right)^{1/n}.$$
(35)

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140

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With constant viscosity model:

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{v + v_{\text{t}}}{v} \left(\frac{D}{\delta}\right)^{1/n} \frac{1}{D} - 28.2$$
$$= \frac{101}{D^{1-1/n} \delta^{1/n}} - 28.2.$$
(36)

With Xu's zero-equation model:

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{\nu + \nu_{\text{t}}}{\nu} \left(\frac{D}{\delta}\right)^{1/n} \frac{1}{D} - 28.2$$
$$\approx \frac{0.03874U_{\infty}}{\nu} \left(\frac{D}{\delta}\right)^{2/n} - 28.2$$
$$= 258 \left(\frac{D}{\delta}\right)^{2/n} - 28.2. \tag{37}$$

With Prandtl's zero-equation model:

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \frac{v + v_{\text{t}}}{v} \left(\frac{D}{\delta}\right)^{1/n} \frac{1}{D} - 28.2$$
$$\approx \frac{2\kappa^2 U_{\infty}}{v} \left(\frac{D}{\delta}\right)^{2/n} - 28.2$$
$$= 2241 \left(\frac{D}{\delta}\right)^{2/n} - 28.2. \tag{38}$$

Eqs. (30)–(32) and (36)–(38) can be illustrated as Fig. 3 by using the common velocity profile of n = 7 and $\delta = 0.13$ m. Fig. 3 indicates that:

- (1) The use of small first grid with the constant viscosity turbulence model would increase the error in predicting the heat flux. A large first grid seems better than a small one.
- (2) The error in predicting the heat flux increases as the grid resolution decreases when the other two zero-equation models are used. A finer grid solution is preferred with the two zero-equation models.

Further analysis of Eq. (37) indicates that $q_{\text{CFD}} = q_{\text{surface}}$ only when $D/\delta = 0.0004$. In fact, under this condition, the first grid falls into the viscous sub-layer (δ_{sub} -layer/ $\delta = 0.15-0.2$). In other words, the first grid should be placed in the sub-layer so as to obtain the correct heat flux. Since $v_t \ll v$ in the sub-layer, Eq. (35) can be re-arranged

$$q_{\rm CFD} = -k \frac{v + v_{\rm t}}{v} \frac{T_{\infty} - T_{\rm surface}}{D} \left(\frac{D}{\delta}\right)^{1/n}$$
$$\approx -k \frac{T_{\infty} - T_{\rm surface}}{D} \left(\frac{D}{\delta}\right)^{1/n}, \tag{39}$$

$$\frac{\Delta q}{k(T_{\text{surface}} - T_{\infty})} = \left(\frac{D}{\delta}\right)^{1/n} \frac{1}{D} - 28.2.$$
(40)



Constant Viscosity Model

Fig. 3. Impact of turbulence model and first grid size on the error of convective heat transfer predicted by CFD.

To achieve $\Delta q = 0, D$ should be 0.03 m if n = 7 and $\delta = 0.13$ m. The value is obtained based on the simple plate flow. Real indoor airflows are more complicated. Usually, the momentum and thermal boundary layer in room airflows are thicker than those in the plate flow. The simulation experience from the literatures (e.g. [14]) indicates that D = 0.05 m is a good value for most indoor airflows. The following section attempts to validate this value through two case studies by using Xu's zero-equation turbulence model.

5. Numerical experiments for two typical room airflows

5.1. Natural convection in a room with an aspect ratio of 2.5:7.9

In order to define a reasonable CFD grid for room airflows, this study first models a natural convection flow in



Fig. 4. Configuration of the experiment [15].

a full-size room as shown in Fig. 4. The room was with a hot wall on one end and a cold wall on its opposite. The room was used by Olson et al. [15] to measure flow and heat transfer. Olson also used a 1:5.5 scale physical model containing R114 gas. The small-scale model was geometrically similar, had the same Reyleigh number, and had the same dimensionless side wall temperature as the full-scale room. The measurement found a good agreement between the full-scale room and the scale model in flow patterns, velocity, temperature distributions, and heat transfer. This study is particularly interested in the convective heat transfer at the hot and cold walls and compares both full-scale and model experimental results with the computed results.

Fig. 5 presents the convective heat transfer at the hot and cold walls for both the measurement and simulation where Nusselt number is plotted as a function of Reyleigh number. Also included in the figure is the correlation from Bohn et al. [16], Nu = $0.31 \text{Ra}^{1/4}$, for enclosure flows. The Nu and Ra number are based on the temperature difference between the hot and cold walls. The experimental uncertainty is approximately 10% for the scale model and 30% for the full-scale room. The results show that the simulations with the first grid size of 0.002-0.005 m agree very well with the data, exhibiting the expected trend of increasing Nu with Ra. The other first grid sizes produce the solutions that



Fig. 6. Influence of first CFD grid size on the calculation of Nu at $Ra = 2.6 \times 10^{10}$.

are more deviated from the measurements, as illustrated in Fig. 6. Note that finer grids do not give more accurate results with the zero-equation model as analyzed. Figs. 5 and 6 indicate that the first grid size at 0.005 m is reasonably good for the indoor natural convection cases.

5.2. Forced convection in a room with an aspect ratio of 5.5:3.7:3.4

The study further investigates a forced convection flow in an experimental chamber with the side-wall jet [17]. The configuration of the experimental facility is shown in Fig. 7. The simulation uses three different grid densities: fine grid $(66 \times 51 \times 45 = 151, 470 \text{ cells})$ has the first grid size at 0.05 m to wall; moderate grid $(44 \times 34 \times 30 = 44, 880 \text{ cells})$ has the first grid at 0.1 m; and coarse grid $(22 \times 17 \times 15 = 5610 \text{ cells})$ has the first grid at 0.2 m. The area-averaged heat fluxes at the room enclosures are then calculated and compared with the measured data, as presented in Fig. 8.

The results show that the simulation with the first grid size at 0.05–0.1 m can provide reasonable solutions for such a forced-convection flow. This conclusion confirms that by Chen [14] who indicated that the first grid size at 0.1 m is a good value for most indoor airflows.



Fig. 5. Comparison of simulated heat transfer at enclosures with the data from [15].



Fig. 7. Schematic of experimental facility [17].

ONTROL



Fig. 8. Comparison of simulated heat flux at enclosures with the measurement.

6. Negative *h* versus convergence and stability of ES

ES and CFD programs can exchange the convective heat information on envelope interior surfaces through different methods. Zhai and Chen [18] enumerated the possible data exchange methods and comparatively studied each of them through theoretical analysis and numerical experiment. According to the study, the fastest and most robust method to exchange convective heat between ES and CFD is that ES provides interior surface temperatures in a room to CFD while CFD returns the convective heat transfer coefficients h and the air temperatures T_D near the surfaces to ES. In order to minimize the modifications in ES programs that use the traditional definition of convection coefficient h based on the temperature difference of an interior surface and room air, a corrected convective heat transfer coefficients, $h_{\text{corrected}}$, rather than h and T_D , can be calculated from CFD results and used in ES. The $h_{\text{corrected}}$ is calculated in CFD through:

$$h_{\text{corrected}} = hA(T_{\text{surface}} - T_D)/A(T_{\text{surface}} - T_{\text{room}}).$$
(41)



Fig. 9. Illustration of negative *h*.

The *h*, calculated in CFD based on the flow viscosity (Eq. (2)), is always positive. However, $h_{\text{corrected}}$ can be negative in some particular cases.

Fig. 9 illustrates such an example in a room with displacement ventilation. If assuming $h = 4 \text{ W/m}^{2\circ}\text{C}$, at the floor surface, the heat gain from the floor $Q = h(T_{\text{floor}} - T_{\text{air}}) = 4 \text{ W/m}^2$. If Q is represented by the temperature difference between T_{control} and T_{floor} , then $Q = h_{\text{corrected}}(T_{\text{floor}} - T_{\text{control}}) = h_{\text{corrected}}(20-24) = 4 \text{ W/m}^2$, one would obtain $h_{\text{corrected}} = -1 \text{ W/m}^{2\circ}\text{C}$. It may even cause the singularity problem if $T_{\text{control}} = 20^{\circ}\text{C}$.

Negative h may cause divergence and instability of an ES simulation. ES solves the following matrix equation for surface energy balance:

$$\mathbf{H} \cdot \mathbf{T} = \mathbf{q},\tag{42}$$

where

$$\mathbf{H} = \begin{bmatrix} h_{1,c} + \sum_{k=1}^{N} h_{1k,r} & -h_{12,r} & \cdots & -h_{1N,r} \\ -h_{21,r} & h_{2,c} + \sum_{k=1}^{N} h_{2k,r} & \cdots & -h_{2N,r} \\ \cdots & \cdots & \cdots & \cdots \\ -h_{N1,r} & \cdots & -h_{NN-1,r} & h_{N,c} + \sum_{k=1}^{N} h_{Nk,r} \end{bmatrix},$$
(43)

$$\mathbf{T} = \begin{bmatrix} T_1 \\ T_2 \\ \cdots \\ T_N \end{bmatrix}, \qquad (44)$$
$$\mathbf{q} = \begin{bmatrix} q_{1,\text{in}} + h_{1,c}T_{\text{room}} \\ q_{2,\text{in}} + h_{2,c}T_{\text{room}} \\ \cdots \\ q_{N,\text{in}} + h_{N,c}T_{\text{room}} \end{bmatrix}$$

and T_i is the temperature of interior surface i, $h_{i,c}$ is the convective heat transfer coefficient of surface *i*, $h_{ii,r}$ is the radiative heat transfer coefficient between surface i and surface $j, q_{i,in}$ is incoming heat to surface *i* (e.g. the conductive heat through the envelopes), N is the total surface number. Therefore,

$$\mathbf{T} = \mathbf{q}/\mathbf{H}.\tag{46}$$

From matrix theory [19], T has a unique solution if only if $|\mathbf{H}| \neq 0$, i.e. **H** is nonsingular. **H** is singular if and only if the rank of $n \times n$ matrix $\mathbf{H} < n$, which means that at least one row in H could be represented by the algebraic combination of the others. Due to the randomness of the coefficients in H, it is impossible to anticipate the determinant of matrix H in general. However, the energy equation for each surface (each row in Eq. (42)), although connected with other surfaces, cannot be determined by energy balances of the other surfaces. Therefore, matrix **H** is nonsingular, regardless of the sign of h.

When iteratively solving Eq. (42) in ES, one may still meet the instability and divergence problems. Matrix theory [19] proves that the matrix should possess some properties to guarantee a converged solution. Following is a brief discussion on the issue with Jacobi method.

Eq. (42) can be rewritten in the following manner:

$$\begin{bmatrix} 1 & h_{1,2}/h_{1,1} & \cdots & h_{1,N}/h_{1,1} \\ h_{2,1}/h_{2,2} & 1 & \cdots & h_{2,N}/h_{2,2} \\ \cdots & \cdots & \cdots & \cdots \\ h_{N,1}/h_{N,N} & \cdots & h_{N,N-1}/h_{N,N} & 1 \end{bmatrix} \begin{bmatrix} T_1 \\ T_2 \\ \cdots \\ T_N \end{bmatrix}$$

$$= \begin{bmatrix} q_1/h_{1,1} \\ q_2/h_{2,2} \\ \cdots \\ q_N/h_{N,N} \end{bmatrix}, \qquad (47)$$

$$\begin{bmatrix} T_1 \\ T_2 \\ \cdots \\ q_N/h_{N,N} \end{bmatrix} = \begin{bmatrix} q_1/h_{1,1} \\ q_2/h_{2,2} \\ \cdots \\ q_N/h_{N,N} \end{bmatrix}$$

$$- \begin{bmatrix} 0 & h_{1,2}/h_{1,1} & \cdots & h_{1,N}/h_{1,1} \\ h_{2,1}/h_{2,2} & 0 & \cdots & h_{2,N}/h_{2,2} \\ \cdots \\ h_{N,1}/h_{N,N} & \cdots & h_{N,N-1}/h_{N,N} & 0 \end{bmatrix} \begin{bmatrix} T_1 \\ T_2 \\ \cdots \\ T_N \end{bmatrix}. \qquad (48)$$

By iteratively solving (48), one can obtain the solution with a prescribed accuracy. Assuming after the *m*th iteration,

$$\mathbf{T}_m = \mathbf{q}' - \mathbf{H}' \mathbf{T}_{m-1}.$$
 (49)

And if T is the exact solution, i.e.

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$$\mathbf{T} = \mathbf{q}' - \mathbf{H}'\mathbf{T}.\tag{50}$$

It can be easily seen, upon subtraction, that

$$\mathbf{T}_{m} - \mathbf{T} = -\mathbf{H}'(\mathbf{T}_{m-1} - \mathbf{T}) = {\mathbf{H}'}^{2}(\mathbf{T}_{m-2} - \mathbf{T})$$

= $(-1)^{m} {\mathbf{H}'}^{m}(\mathbf{T}_{0} - \mathbf{T}).$ (51)

Hence,

$$\lim_{m \to \infty} (T_m - T) = 0 \quad \text{if } \lim_{m \to \infty} \mathbf{H}'^m = 0.$$
 (52)

In other words, a necessary and sufficient condition for the convergence of the Jacobi method is that \mathbf{H}^{m} tends to zero as *m* tends to infinite. Such a limit occurs if the spectral radius of H is less than unit. For the moment, a sufficient condition can be

$$|\mathbf{H}| < 1 \tag{53}$$

since

$$|\mathbf{H}^m| \leqslant |\mathbf{H}|^m. \tag{54}$$

From matrix theory, a sufficient condition when $|\mathbf{H}|$ is less than unity is to satisfy the following condition:

$$\sum_{\substack{j=1\\i\neq i}}^{N} \left| \frac{h_{i,j}}{h_{i,i}} \right| < 1, \quad i = 1, 2, \dots, N.$$
(55)

With positive *h* values, the elements in the matrix **H** of Eq. (42) always satisfy

$$\left| h_{i,c} + \sum_{j=1}^{N} h_{ij,r} \right| > \sum_{\substack{j=1\\ j \neq i}}^{N} |-h_{ij,r}|$$
(56)

which makes H a diagonal dominant matrix, assuring that there exists a unique solution for the vector **T**.

When $h_{i,c}$ is negative, Eqs. (55) or (56) may not be satisfied, which could cause divergence and instability during a calculation. The divergence and instability may not always occur, since satisfying Eq. (55) is only a sufficient condition. In general, the larger the real situation departs from Eq. (55) (i.e. the larger negative $h_{i,c}$), the higher the probability of divergence and instability.

7. Conclusions

In a coupled simulation of ES and CFD, CFD provides convective heat transfer on interior surfaces of building envelope to ES. Accurate prediction of the convective heat transfer by CFD is crucial to the accuracy of building energy calculation. The size of first grid to an interior surface and turbulence model in CFD are most important to the calculation of convective heat transfer. The convective heat transfer determined from the first grid cell, rather than from a prescribed location, should be used in the coupled simulation.

This investigation found that the calculated convective heat transfer depends on the size of first grids by analyzing laminar flows of forced and natural convection over a flat plate. The error of convective heat transfer due to the size of first grids (2D) is on the order of $O(D^2)$ for laminar forced convection and O(D) for natural convection.

This study further discussed the combined effect of the size of first grids and turbulence model on the convective heat transfer. Three zero-equation models have been used: the constant viscosity model, Xu's model, and the Prandtl mixing length model. A finer grid distribution in CFD does not always lead to a more accurate solution when the constant viscosity turbulence model is used or when the first grid is in the sub-layer boundary with the other zero-equation turbulence models. Based on the theoretical analysis and numerical experimentation, the study suggests a universal first grid size of 0.005 m for indoor natural convection airflows and 0.1m for indoor forced convection airflows.

This investigation also found that the convective heat transfer coefficients, using traditional definition for building energy simulation, could become negative for room with strong temperature stratification, such as with displacement ventilation. The theoretical analysis shows that a negative convective heat transfer coefficient may cause divergence and instability in energy simulation.

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